## 1. Atmospheric eddy

- (a) Radial velocity  $v_c = 10 \,\mathrm{ms}^{-1}$  at R, radius  $R = 500 \,\mathrm{km}$ 
  - Circulation around a circle:

$$C = \int_0^{2\pi} v_c d_c = 2\pi \cdot R \cdot v_c$$
$$= 2\pi \cdot 5 \cdot 10^5 \text{m} \cdot 10 \text{ ms}^{-1}$$
$$= \pi \cdot \mathbf{10^7 m^2 s^{-1}}$$

• Mean vorticity:

$$\zeta = \frac{C}{\pi R^2} = \frac{2\pi R v_c}{\pi R^2} = \frac{2v_c}{R}$$
$$= \mathbf{4} \cdot \mathbf{10^{-5} s^{-1}}$$

(b) Vorticity in polar coordinates (see lecture notes page ??):

$$\zeta = \frac{1}{r} \frac{\partial}{\partial r} (rv) = \frac{v}{r} + \frac{\partial v}{\partial r}$$

Assuming that there is no vorticity in the region outside R, leads to:

$$\zeta = \begin{cases} \frac{2 v_c}{R} & r = R\\ 0 & r > R \end{cases}$$

Derive the expression for v(r) if r > R ( $\xi$  must be 0):

$$\frac{v}{r} + \frac{\partial v}{\partial r} = 0$$
$$\frac{\partial v}{v} = -\frac{\partial r}{r}$$
$$\ln v = -\ln r + K$$
$$v(r) = K \cdot e^{-\ln r}$$
$$v(r) = \frac{K}{r}$$

Here K is a constant. That means that the velocity decreases outside R with K/r!Derive the value of K with the boundary condition  $v = v_c = 10 \text{ ms}^{-1}$  if  $r \to R$ .

$$K = v_c \cdot R$$
$$= 5 \cdot 10^6 \,\mathrm{m^2 s^{-1}}$$

Now we can calculate the the velocity at radius  $1000\,{\rm km}$  entierly induced by the vorticity within  $500\,{\rm km}$ :

$$v(r) = \frac{K}{r} = \frac{v_c \cdot R}{r}$$
$$v(1000 \text{ km}) = \frac{10 \text{ ms}^{-1} \cdot 5 \cdot 15^5 \text{ m}}{10^6 \text{ m}}$$
$$= 5 \text{ ms}^{-1}$$

Two velocity fields, one from the uniform flow  $v_1$  and one from the two-dimensional vortex  $v_2$ .

•  $v_1$ : The streamfunction  $\psi$  in polar coordinates is given by  $\psi = -Ur \sin \Theta$  (in Cartesian coordinates  $\psi = -U \cdot y$ ), therefore:

$$u_1 = -\frac{1}{r} \frac{\partial \psi}{\partial \Theta} = U \cos \Theta$$
$$v_1 = \frac{\partial \psi}{\partial r} = -U \sin \Theta$$

• v<sub>2</sub>:

$$v_2 = \frac{C}{2\pi a} = \frac{\Lambda \pi a^2}{2\pi a} = \frac{\Lambda a}{2}$$

Vortex in stationary flow:

$$v = v_1 + v_2 = \begin{pmatrix} U\cos\Theta\\ -U\sin\Theta + \frac{\Lambda a}{2} \end{pmatrix}$$

The stagnation point is located where v = 0  $(U = \frac{\Lambda a}{2})$ :

$$v = \frac{\Lambda a}{2} \begin{pmatrix} \cos \Theta \\ \sin \Theta + 1 \end{pmatrix} = 0$$

Therefore:

$$\cos \Theta = 0$$
$$\sin \Theta = 1$$

That means that the stagnation point is located at the edge of the vortex (r = a) at  $\Theta = \pi/2$ .

3. Rossby and Ekman numbers

Rossby number:  $R_0 = \frac{U}{fL}$   $\left(\frac{\text{velocity}}{\text{coriolis}}\right)$ 

Ekman number:  $E = \frac{\nu}{fL^2}$   $\left(\frac{\text{friction}}{\text{coriolis}}\right)$ 

	L [m]	$U  [\mathrm{ms}^{-1}]$	$\nu  [\mathrm{m^2 s^{-1}}]$	$R_0$	E
air	$10^{6}$	$10^{1}$	$1.5 \cdot 10^{-5}$	$10^{-1}$	$1.5 \cdot 10^{-13}$
water	$10^{5}$	$10^{-2}$	$1.0 \cdot 10^{-6}$	$10^{-3}$	$1.0 \cdot 10^{-12}$

 $R_0$  and  $E \ll 1$  for atmospheric and oceanic flows. That means the inertial (U) and friction  $(\nu)$  effects are small compared with the Coriolis force.

Remark:  $R_{0_{air}}$  is larger than  $R_{0_{water}}$ . The inertial effects are more important in the atmosphere than int the ocean.

## 4. Geostrophic approximation

Navier-Stokes equation (see lecture notes page 101):

$$R_0 \frac{Dv}{Dt} - \left(\vec{k} \times \vec{v}\right) f = -\frac{1}{\rho} \vec{\nabla} p + E \nabla^2 \nu$$

With  $R_0 =$ Rossby number and E =Ekman number.

For large scale flows (synoptic scale) the Rossby- and Ekman numbers are << 1 as seen in the solution above. Therefore term (1) and (4) can be neclected. A geostrophic balance exists between the Coriolis- and the pressure gradient force.

$$fv = \frac{1}{\rho} \frac{\partial p}{\partial x}$$
$$fu = -\frac{1}{\rho} \frac{\partial p}{\partial y}$$

If we don't split the equation in a meridional and zonal component we get:

$$f\vec{v} = \frac{1}{\rho}\frac{\partial p}{\partial \tilde{x}}$$

Geostrophic velocity:

$$\vec{v} = \frac{1}{\rho f} \frac{\partial p}{\partial \tilde{x}}$$
$$= \frac{1}{1 \text{ kg m}^{-3} \cdot 10^{-4}} \frac{10^3 \text{ Pa}}{10^6 \text{ m}}$$
$$\sim \mathbf{10 \text{ ms}^{-1}}$$